

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

Turbine Kinetics, Inc.  
 A HEICO Aerospace Company  
 60 Sequin Drive  
 Glastonbury, CT 06033

PMA NO. PQ0816NE  
 SUPPLEMENT NO. 557  
 DATE April 25, 2007

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Guide, JT8D No. 4 & 5 Bearing Inner Heat Shield – Scavenge Line	PT010082	SR810279	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> PT010082 <u>Rev:</u> A <u>Date:</u> 4/25/00 or later FAA approved revisions	Pratt & Whitney	JT8D-1, JT8D-1A, JT8D-1B, JT8D-7, JT8D-7A, JT8D-7B, JT8D-9, JT8D-9A, JT8D-11, JT8D-15, JT8D-15A, JT8D-17, JT8D-17A, JT8D-17R, JT8D-17AR, JT8D-209, J78D-217, JT8D-217A
Guide, JT8D No. 4 & 5 Bearing Inner Heat Shield – Pressure Line	PT010083	SR810280	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> PT010083 <u>Rev:</u> A <u>Date:</u> 4/25/00 or later FAA approved revisions	Pratt & Whitney	JT8D-1, JT8D-1A, JT8D-1B, JT8D-7, JT8D-7A, JT8D-7B, JT8D-9, JT8D-9A, JT8D-11, JT8D-15, JT8D-15A, JT8D-17, JT8D-17A, JT8D-17R, JT8D-17AR, JT8D-209, J78D-217, JT8D-217A
Seal – Instrumentation	PT010138	803510	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> PT010138 <u>Rev:</u> B <u>Date:</u> 3/8/01 or later FAA approved revisions	Pratt & Whitney	JT8D-209, J78D-217, JT8D-217A, JT8D-217C, JT8D-219

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Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Sleeve, Aligning – Compressor Stator Stage IGV – 4	PT030017D05	9646M51P05	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030017 <u>Rev:</u> None <u>Date:</u> 10/3/90 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2  CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6, CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B5F, CF6-80C2B6F, CF6-80C2B6FA, CF6-80C2B7F, CF6-80C2B8F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4

\* Model variations shown in Note 12 of the TCDS

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Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Pin – Shroud Retainer HPT Stator	PT030023D01	9283M71P01	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030023 <u>Rev:</u> A <u>Date:</u> 4/11/94 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4
Pin – Shroud Retainer HPT Stator	PT030023D02	9283M71P02	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030023 <u>Rev:</u> A <u>Date:</u> 4/11/94 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4

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Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Pin – Shroud Retainer HPT Stator	PT030023D03	9283M71P03	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030023 <u>Rev:</u> A <u>Date:</u> 4/11/94 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4
Pin – Shroud Retainer HPT Stator	PT030023D05	9283M71P05	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030023 <u>Rev:</u> A <u>Date:</u> 4/11/94 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4

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Stop, Blade Retainer-Compressor Rotor-Stage 2	PT030026D03	9173M86P03	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030026 <u>Rev:</u> A <u>Date:</u> 5/4/94 or later FAA approved revisions	General Electric	CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2  CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B5F, CF6-80C2B6F, CF6-80C2B6FA, CF6-80C2B7F, CF6-80C2B8F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4
Seal, HP Turb. Stator Outer	PT030032A01	1713M33G01	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030032 <u>Rev:</u> A <u>Date:</u> 5/4/94 or later FAA approved revisions	General Electric	CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R,

\* Model variations shown in Note 12 of the TCDS

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Retainer, Shroud, High Pressure Turb Stator Stg 2	PT030034A01	9374M75G01	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030034 <u>Rev:</u> B <u>Date:</u> 3/21/03 or later FAA approved revisions	General Electric	CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6, CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F, CF6-80E1A2
Retainer, Shroud, High Pressure Turb Stator Stg 2	PT030034A03	9374M75G03	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030034 <u>Rev:</u> B <u>Date:</u> 3/21/03 or later FAA approved revisions	General Electric	CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6, CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F
Retainer, Shroud, High Pressure Turb Stator Stg 2	PT030034A04	9374M75G04	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> PT030034 <u>Rev:</u> B <u>Date:</u> 3/21/03 or later FAA approved revisions	General Electric	CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6, CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B5F, CF6-80C2B6F, CF6-80C2B6FA, CF6-80C2B7F, CF6-80C2B8F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4

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Liner, Vane Dovetail – Compressor Stator	PT030035D07	9073M97P07	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2
Liner, Vane Dovetail – Compressor Stator	PT030035D08	9073M97P08	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2
Liner, Vane Dovetail – Compressor Stator	PT030035D09	9073M97P09	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2
Liner, Vane Dovetail – Compressor Stator	PT030035D10	9073M97P10	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2

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Liner, Vane Dovetail – Compressor Stator	PT030035D11	9073M97P11	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2  CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F

\* Model variations shown in Note 12 of the TCDS

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Liner, Vane Dovetail – Compressor Stator	PT030035D12	9073M97P12	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2  CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6, CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F
Liner, Vane Dovetail – Compressor Stator	PT030035D13	9073M97P13	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2

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Liner, Vane Dovetail – Compressor Stator	PT030035D14	9073M97P14	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2
Liner, Vane Dovetail – Compressor Stator	PT030035D19	9073M97P19	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3
Liner, Vane Dovetail – Compressor Stator	PT030035D20	9073M97P20	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3
Liner, Vane Dovetail – Compressor Stator	PT030035D21	9073M97P21	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3

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Liner, Vane Dovetail – Compressor Stator	PT030035D22	9073M97P22	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3
Liner, Vane Dovetail – Compressor Stator	PT030035D29	9073M97P29	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F
Liner, Vane Dovetail – Compressor Stator	PT030035D30	9073M97P30	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030035 <u>Rev:</u> B <u>Date:</u> 4/30/99 or later FAA approved revisions	General Electric	CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B6F, CF6-80C2A5F
Insulator End Piece	PT030039A01	1647M74G01	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030039 <u>Rev:</u> None <u>Date:</u> 3/10/97 or later FAA approved revisions	General Electric	CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2

\* Model variations shown in Note 12 of the TCDS

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Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Retainer, Blade, Comp. Rotor, STG 1	PT030042D01	9181M55P01	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030042 <u>Rev:</u> None <u>Date:</u> 9/21/92 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2  CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6
Retainer, Blade, Comp. Rotor, STG 1	PT030042D02	9181M55P02	Identicality per 14 CFR § 21.303 <u>DWG No:</u> PT030042 <u>Rev:</u> None <u>Date:</u> 9/21/92 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2  CF6-80A, CF6-80A1, CF6-80A2, CF6-80A3  CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6, CF6-80C2D1F

\* Model variations shown in Note 12 of the TCDS

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FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

Turbine Kinetics, Inc.  
A HEICO Aerospace Company  
60 Sequin Drive  
Glastonbury, CT 06033

PMA NO. PQ0816NE  
SUPPLEMENT NO. 557  
DATE April 25, 2007

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Retainer, Blade - LP Turb. Rotor Stg. 4 and 5	PT030053D01	9382M63P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030053 <u>Rev:</u> None <u>Date:</u> 8/13/99 or later FAA approved revisions	General Electric	CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B5F, CF6-80C2B6F, CF6-80C2B6FA, CF6-80C2B7F, CF6-80C2B8F, CF6-80C2A5F  CF6-80E1A2, CF6-80E1A3, CF6-80E1A4
Retainer, Blade - LP Turb. Rotor Stg. 4 and 5	PT030053D02	9382M63P02	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030053 <u>Rev:</u> None <u>Date:</u> 8/13/99 or later FAA approved revisions	General Electric	CF6-80C2A1, CF6-80C2A2, CF6-80C2A3, CF6-80C2A5, CF6-80C2A8, CF6-80C2B1, CF6-80C2B2, CF6-80C2B4, CF6-80C2B6 , CF6-80C2D1F, CF6-80C2B1F, CF6-80C2B2F, CF6-80C2B4F, CF6-80C2B5F, CF6-80C2B6F, CF6-80C2B6FA, CF6-80C2B7F, CF6-80C2B8F, CF6-80C2A5F
Flange, Nozzle Support – Turbine Mid Frame	PT030057D04	3049M88P04	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT030057 <u>Rev:</u> None <u>Date:</u> 10/16/98 or later FAA approved revisions	General Electric	CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2

\* Model variations shown in Note 12 of the TCDS

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FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

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Glastonbury, CT 06033

PMA NO. PQ0816NE  
SUPPLEMENT NO. 557  
DATE April 25, 2007

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Nozzle Flange	PT060017D01	1755M32P01	Test & Computations per 14 CFR § 21.303 <u>DWG No:</u> PT060017, <u>Rev:</u> A <u>Date:</u> 3/23/01 or later FAA approved revisions  <u>DWG No:</u> PT060017CAST, <u>Rev:</u> A <u>Date:</u> 4/9/01 or later FAA approved revisions	CFM International	CFM56-2, CFM56 -2A, CFM56-2B, CFM56-3, CFM56-3B, CFM56-3C, CFM56-5, CFM56-5A1/F, CFM56-5A3, CFM56-5A4, CFM56-5A4/F, CFM56-5A5, CFM56-5A5/F, CFM56-5B1, CFM56-5B2, CFM56-5B4, CFM56-5B5, CFM56-5B6, CFM56-5B7, CFM56 -5B1/P, CFM56-5B2/P, CFM56-5B3/P, CFM56-5B4/P, CFM56-5B5/P, CFM56-5B6/P, CFM56-5B7/P, CFM56-5C2, CFM56-5C2/F, CFM56-5C3/F, CFM56-5C2/G, CFM56-5C3/G, CFM56-5C2/4, CFM56-5C2/F4, CFM56-5C2/G4, CFM56-5C3/F4, CFM56-5C3/G4, CFM56-5C4, CFM56-5C4/1, CFM56-7B18, CFM56-7B20, CFM56-7B22, CFM56-7B24, CFM56-7B26, CFM56-7B27, CFM56-7B22/B1, CFM56-7B24/B1, CFM56-7B26/B1, CFM56-7B27/B1, CFM56-7B27/B3, CFM56-7B22/B2, CFM56-7B26/B2

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FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

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Glastonbury, CT 06033

PMA NO. PQ0816NE  
SUPPLEMENT NO. 557  
DATE April 25, 2007

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Brush Seal	PT120009A01	50J839-01	Test & Computations per 14 CFR § 21.303 <u>DWG No:</u> PT120009, <u>Rev:</u> A <u>Date:</u> 5/7/01 or later FAA approved revisions	Pratt & Whitney	PW4052*, PW4056*, PW4060*, PW4062*, PW4152*, PW4156A*, PW4158*, PW4460*, PW4462*, PW4164, PW4168, PW4168A, PW4074, PW4077, PW4074D, PW4077D, PW4084D, PW4090, PW4090-3, PW4090D, PW4098
Fuel Nozzle Guide	PT120011	50J988	Test & Computations per 14 CFR § 21.303 <u>DWG No:</u> PT120011, <u>Rev:</u> A <u>Date:</u> 3/20/06 or later FAA approved revisions	Pratt & Whitney	PW4074, PW4077
Fuel Nozzle Stator	PT120012	50J781	Test & Computations per 14 CFR § 21.303 <u>DWG No:</u> PT120012, <u>Rev:</u> None <u>Date:</u> 3/17/04 or later FAA approved revisions	Pratt & Whitney	PW4052*, PW4056*, PW4060*, PW4062*, PW4152*, PW4156A*, PW4158*, PW4460*, PW4462*

\*Includes derivatives identified as (-3)

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FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

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A HEICO Aerospace Company  
60 Sequin Drive  
Glastonbury, CT 06033

PMA NO. PQ0816NE  
SUPPLEMENT NO. 557  
DATE April 25, 2007

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Fuel Nozzle Guide	PT120013	50J873	Test & Computations per 14 CFR § 21.303 <u>DWG No:</u> PT120013, <u>Rev:</u> A <u>Date:</u> 3/21/06 or later FAA approved revisions	Pratt & Whitney	PW4074D, PW4077D, PW4084D, PW4090, PW4090-3
Runner Assy – Synch Ring	PT120050A01	55H550	Test & Computations per 14 CFR § 21.303 <u>DWG No:</u> PT120050, <u>Rev:</u> A <u>Date:</u> 7/30/02 or later FAA approved revisions	Pratt & Whitney	PW4052*, PW4056*, PW4060*, PW4060C, PW4062*, PW4152*, PW4156A*, PW4158*, PW4460*, PW4462*
*Includes derivatives identified as (-3)					
Seal, Aft Outer Liner Turbine Mid Frame	PT6553D01	9084M71P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT6553 <u>Rev:</u> A <u>Date:</u> 4/28/94 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2

\* Model variations shown in Note 12 of the TCDS

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FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

Turbine Kinetics, Inc.  
 A HEICO Aerospace Company  
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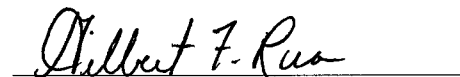
PMA NO. PQ0816NE  
 SUPPLEMENT NO. 557  
 DATE April 25, 2007

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Seal, Aft Outer Liner Turbine Mid Frame	PT6553D02	9084M71P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> PT6553 <u>Rev:</u> A <u>Date:</u> 4/28/94 or later FAA approved revisions	General Electric	CF6-6D, CF6-6D1, CF6-6D1A, CF6-6K, CF6-6K2  CF6-45A, CF6-45A2, CF6-50CA, CF6-50C, CF5-50C1, CF6-50C2, CF6-50C2B, CF6-50C2D, *CF6-50C2-F, *CF6-50C2-R, CF6-50E, CF6-50E1, CF6-50E2
* Model variations shown in Note 12 of the TCDS					
HPT 1 <sup>st</sup> Stage Damper	1B8049PT	1B8049	Test & Computations per 14 CFR § 21.303 <u>DWG No:</u> 1B8049PT, <u>Rev:</u> A <u>Date:</u> 3/3/06 or later FAA approved revisions	Pratt & Whitney	PW2037, PW2037M, PW2040, PW2240, PW2337

----- (END OF LISTING) -----

NOTE: Provide minor design changes in a manner as determined by the ACO. Handle major design changes to drawings and specifications in the same manner as that for an original FAA-PMA. If TC holder's ICA applies to these replacement parts, provide a statement noting such. If not, provide supplementary ICA per 14 CFR § 21.50.

  
 Wego Wang  
 Aerospace Engineer  
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 Gilbert F. Rua  
 Manager, Manufacturing Inspection  
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