

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

Turbine Kinetics, Inc.
A HEICO Aerospace Company
60 Sequin Drive
Glastonbury, CT 06033

PMA NO. PQ0816NE
SUPPLEMENT NO. 235 Amdt 1
DATE June 22, 2011

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Guide	9511M38P01QQ	9511M38P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 359 <u>Rev:</u> None <u>Date:</u> 2/19/99 or later FAA approved revisions	CFM International	CFM56-2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1
	9511M38P02QQ	9511M38P02			
	9511M38P09QQ	9511M38P09			
	9511M38P10QQ	9511M38P10			
	9511M38P11QQ	9511M38P11			
	9511M38P12QQ	9511M38P12			
	9511M38P13QQ	9511M38P13			
	9511M38P14QQ	9511M38P14			
	9511M38P15QQ	9511M38P15			
9511M38P16QQ	9511M38P16				
Guide	9511M38P03QQ	9511M38P03	Identity per 14 CFR § 21.303 <u>DWG No:</u> 359 <u>Rev:</u> None <u>Date:</u> 2/19/99 or later FAA approved revisions	CFM International	CFM56-2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1
	9511M38P04QQ	9511M38P04			
	9511M38P05QQ	9511M38P05			
	9511M38P06QQ	9511M38P06			
	9511M38P07QQ	9511M38P07			
	9511M38P08QQ	9511M38P08			
Guide	9511M39P01QQ	9511M39P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 213 <u>Rev:</u> A <u>Date:</u> 7/8/99 or later FAA approved revisions	CFM International	CFM56-2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1
	9511M39P02QQ	9511M39P02			
Guide, Sector, Compressor Stator	9511M40P01QQ	9511M40P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 184 <u>Rev:</u> A <u>Date:</u> 6/11/96 or later FAA approved revisions	CFM International	CFM56-2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -2A, -2A-2, -2A-3, -2B, -2B-1, -3, -3-B1, -3B-2, -3C-1
	9511M40P02QQ	9511M40P02			
	9511M40P03QQ	9511M40P03			
	9511M40P04QQ	9511M40P04			
	9511M40P05QQ	9511M40P05			
	9511M40P06QQ	9511M40P06			
	9511M40P07QQ	9511M40P07			
	9511M40P08QQ	9511M40P08			

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Seal, Honeycomb, Stationary High Pressure Turbine, Inner and Outer	9512M40P01QQ 9512M40P02QQ 9512M40P03QQ 9512M40P04QQ 9512M40P05QQ 9512M40P06QQ 9512M40P08QQ 9512M40P09QQ	9512M40P01 9512M40P02 9512M40P03 9512M40P04 9512M40P05 9512M40P06 9512M40P08 9512M40P09	Identicity per 14 CFR § 21.303 <u>DWG No:</u> 195 <u>Rev:</u> A <u>Date:</u> 8/16/96 or later FAA approved revisions	CFM International	CFM56-2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -2B, -2B-1, -3, -3-B1, -3B-2, -3C-1

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Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Pin, Threaded	9513M98P01QQ	9513M98P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 200 <u>Rev:</u> None <u>Date:</u> 7/24/96 or later FAA approved revisions	CFM International	CFM56-2, CFM56-2A, CFM56-2B, CFM56-3, CFM56-3B, CFM56-3C, CFM56-5, CFM56-5-A1, CFM56-5A1/F, CFM56-5A3, CFM56-5A4, CFM56-5A4/F, CFM56-5A5, CFM56-5A5/F, CFM56-5B1, CFM56-5B1/2, CFM56-5B1/2P, CFM56-5B1/P, CFM56-5B2, CFM56-5B2/2, CFM56-5B2/2P, CFM56-5B2/P, CFM56-5B3/2P, CFM56-5B3/P, CFM56-5B4, CFM56-5B4/2, CFM56-5B4/2P, CFM56-5B4/P, CFM56-5B5, CFM56-5B5/P, CFM56-5B6, CFM56-5B6/2, CFM56-5B6/2P, CFM56-5B6/P, CFM56-5C2, CFM56-5C2/4, CFM56-5C2/F, CFM56-5C2/F4, CFM56-5C2/G, CFM56-5C2/G4, CFM56-5C3/F, CFM56-5C3/F4, CFM56-5C3/G CFM56-5C3/G4, CFM56-5C4, CFM56-5C4/1, CFM56-7B20, CFM56-7B20/2, CFM56-7B20/3, CFM56-7B22, CFM56-7B22/2, CFM56-7B22/3, CFM56-7B22/3B1, CFM56-7B22/B1, CFM56-7B24, CFM56-7B24/2, CFM56-7B24/3, CFM56-7B24/3B1, CFM56-7B24/B1, CFM56-7B26, CFM56-7B26/2, CFM56-7B26/3, CFM56-7B26/3B1, CFM56-7B26/3B2, CFM56-7B26/3B2F, CFM56-7B26/3F, CFM56-7B26/B1, CFM56-7B26/B2, CFM56-7B27, CFM56-7B27/2, CFM56-7B27/3, CFM56-7B27/3B1, CFM56-7B27/3B1F, CFM56-7B27/3B3, CFM56-7B27/3F, CFM56-7B27/B1, CFM56-7B27/B3, CFM56-7B27A, CFM56-7B27A/3

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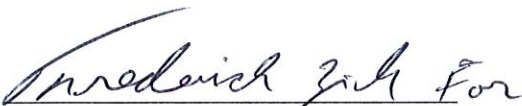
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Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Bolt, Machine, Double Hex Head	9514M84P01QQ	9514M84P01	Identicality per 14 CFR § 21.303 <u>DWG No: 189</u> <u>Rev: D</u> <u>Date: 11/3/99</u> or later FAA approved revisions	CFM International	CFM56-2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -2A, -2A-2, -2A-3, -2B, -2B-1, -3, -3-B1, -3B-2, -3C-1
Bolt, Slabbed Head	9514M85P01QQ	9514M85P01	Identicality per 14 CFR § 21.303 <u>DWG No: 117</u> <u>Rev: D</u> <u>Date: 7/6/00</u> or later FAA approved revisions	CFM International	CFM56-2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1

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NOTE:

- 1) Provide minor design changes in a manner as determined by the ACO. Handle major design changes to drawings and specifications in the same manner as that for an original FAA-PMA.
- 2) The FAA approved ICA for the above parts with their designs. These ICA may refer to those of the respective parts from the holders of type certificates. Otherwise, provide supplemental ICA for differences in the replacement parts. Make referral statements or supplemental ICA readily available per 14 CFR 21.50.


 Manager,
 Engine Certification Office, ANE-140


 Steven J. Sykes
 Manager, Manufacturing Inspection
 District Office, ANE-MIDO-41