

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

Turbine Kinetics, Inc.
 Subsidiary of HEICO Corp.
 60 Sequin Drive
 Glastonbury, CT 06033

PMA NO. PQ816NE-D
 SUPPLEMENT NO. 235 REV 1
 DATE December 17, 2001

Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Guide	9511M38P01QQ	9511M38P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 359 <u>Rev:</u> None <u>Date:</u> 2/19/99 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1
	9511M38P02QQ	9511M38P02			
	9511M38P09QQ	9511M38P09			
	9511M38P10QQ	9511M38P10			
	9511M38P11QQ	9511M38P11			
	9511M38P12QQ	9511M38P12			
	9511M38P13QQ	9511M38P13			
	9511M38P14QQ	9511M38P14			
	9511M38P15QQ	9511M38P15			
	9511M38P16QQ	9511M38P16			
Guide	9511M38P03QQ	9511M38P03	Identity per 14 CFR § 21.303 <u>DWG No:</u> 359 <u>Rev:</u> None <u>Date:</u> 2/19/99 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1
	9511M38P04QQ	9511M38P04			
	9511M38P05QQ	9511M38P05			
	9511M38P06QQ	9511M38P06			
	9511M38P07QQ	9511M38P07			
	9511M38P08QQ	9511M38P08			
Guide	9511M39P01QQ	9511M39P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 213 <u>Rev:</u> A <u>Date:</u> 7/8/99 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1
	9511M39P02QQ	9511M39P02			
Guide, Sector, Compressor Stator	9511M40P01QQ	9511M40P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 184 <u>Rev:</u> A <u>Date:</u> 6/11/96 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -2A, -2A-2, -2A-3, -2B, -2B-1, -3, -3-B1, -3B-2, -3C-1
	9511M40P02QQ	9511M40P02			
	9511M40P03QQ	9511M40P03			
	9511M40P04QQ	9511M40P04			
	9511M40P05QQ	9511M40P05			
	9511M40P06QQ	9511M40P06			
	9511M40P07QQ	9511M40P07			
	9511M40P08QQ	9511M40P08			
Seal, Honey- comb, Stationary High Pressure Turbine, Inner and Outer	9512M40P01QQ	9512M40P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 195 <u>Rev:</u> A <u>Date:</u> 8/16/96 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -2B, -2B-1, -3, -3-B1, -3B-2, -3C-1
	9512M40P02QQ	9512M40P02			
	9512M40P03QQ	9512M40P03			
	9512M40P04QQ	9512M40P04			
	9512M40P05QQ	9512M40P05			
	9512M40P06QQ	9512M40P06			
	9512M40P08QQ	9512M40P08			
	9512M40P09QQ	9512M40P09			

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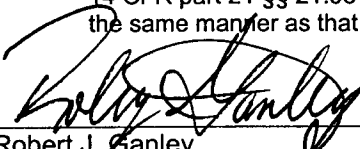
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Part Name	Part Number	Approved Replacement for Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Pin, Threaded	9513M98P01QQ	9513M98P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 200 <u>Rev:</u> None <u>Date:</u> 7/24/96 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -2A, -2A-2, -2A-3, -2B, -2B-1, -3, -3-B1, -3B-2, -3C-1, -5, -5-A1, -5A1/F, -5A3, -5A4, -5A4/F, -5A5, -5A5/F, -5B1, -5B1/2, -5B1/2P, -5B1/P, -5B2, -5B2/2, -5B2/2P, -5B2/P, -5B3/2P, -5B3/P, -5B4, -5B4/2, -5B4/2P, -5B4/P, -5B5, -5B5/P, -5B6, -5B6/2, -5B6/2P, -5B6/P, -5C2, -5C2/4, -5C2/F, -5C2/F4, -5C2/G, -5C2/G4, -5C3/F, -5C3/F4, -5C3/G, -5C3/G4, -5C4, -5C4/1, -7B18, -7B20, -7B20/2, -7B22, -7B22/2, -7B22/B1, -7B24, -7B24/2, -7B24/B1, -7B26, -7B26/2, -7B26/B1, -7B27, -7B27/2, -7B27/B1, -7B27/B3
Bolt, Machine, Double Hex Head	9514M84P01QQ	9514M84P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 189 <u>Rev:</u> D <u>Date:</u> 11/3/99 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -2A, -2A-2, -2A-3, -2B, -2B-1, -3, -3-B1, -3B-2, -3C-1
Bolt, Slabbed Head	9514M85P01QQ	9514M85P01	Identity per 14 CFR § 21.303 <u>DWG No:</u> 117 <u>Rev:</u> D <u>Date:</u> 7/6/00 or later FAA approved revisions.	CFM International	CFM56 -2, -2-C1, -2-C2, -2-C3, -2-C5, -2-C6, -3, -3-B1, -3B-2, -3C-1

----- (END OF LISTING) -----

NOTE: Minor design changes (reference 14 CFR part 21 §§ 21.93 and 21.95) must be submitted in a manner as determined by the ACO. Major design changes (reference 14 CFR part 21 §§ 21.93 and 21.97) to drawings and specifications are to be handled in the same manner as that for an original FAA-PMA.



 Robert J. Ganley
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 Francis A. Favara
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