

FAA – PARTS MANUFACTURER APPROVAL (PMA) SUPPLEMENT

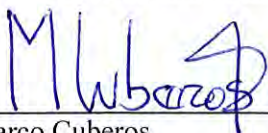
Turbine Kinetics, Inc.  
 A HEICO Aerospace Company  
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 Glastonbury, CT 06033

**PMA NO.:** PQ0816NE  
**SUPPLEMENT No.:** ODA-TK-045  
**DATE:** October 17, 2016

| Article Name     | Article Number  | Approved Replacement for Article Number | Approval Basis and Approved Design Data                                                                                                                          | Make/TCH Eligibility | Model/Series Eligibility                                                                                    |
|------------------|-----------------|-----------------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------|----------------------|-------------------------------------------------------------------------------------------------------------|
| Latch Assy       | 116A8505-1KT    | Hartwell P/N:<br>116A8505-1             | Tests and Computations per 14 CFR § 21.303, Dwg No.: 116A8505-1KT, Rev.: A, Dated: 7/14/2016, or later FAA approved revisions.<br><br><u>NOTE:</u> AMOC Required | Boeing               | 737-700 Series<br>737-800 Series<br>737-600 Series<br>737-700C Series<br>737-900 Series<br>737-900ER Series |
| Seal Disc        | 116A8505-2KT    | Hartwell P/N:<br>116A8505-2             | Tests and Computations per 14 CFR § 21.303, Dwg No.: 116A8505-2KT, Rev.: A, Dated: 8/10/2016, or later FAA approved revisions.<br><br><u>NOTE:</u> AMOC Required | Boeing               | 737-700 Series<br>737-800 Series<br>737-600 Series<br>737-700C Series<br>737-900 Series<br>737-900ER Series |
| Hinge Assy– (RH) | 402D50030-000KT | Boeing P/N:<br>402D50030-000            | Tests and Computations per 14 CFR § 21.303, Dwg No.: 402D50030-000KT, Rev.: – , Dated: 8/18/16, or later FAA approved revisions.                                 | Boeing               | 717-200                                                                                                     |

----- End of Listing -----

NOTE: Minor design changes (reference 14 CFR part 21 §§ 21.319 and 21.619) and major design changes (reference 14 CFR part 21 §§ 21.319 and 21.619) to drawings and specifications must be accomplished in accordance with the FAA approved HEICO PMA ODA Procedures Manual.



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 Marco Cuberos  
 PMA ODA administrator  
 HEICO Aerospace Corporation

OCTOBER 17, 2016

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 Date

This Supplement is an attachment to FAA-PMA approval letter dated March 28, 2001