



PMA Supplements Index:

ARTICLE NUMBER	ARTICLE NAME	PMA SUPP.	DATE ISSUED
DEC47-003-501	GASKET	TK111	FEB 22, 2005
		ODA-TK-019	MARCH 3, 2016

FAA – PARTS MANUFACTURER APPROVAL (PMA) SUPPLEMENT

Turbine Kinetics, Inc.
A HEICO Aerospace Company
60 Sequin Drive
Glastonbury, CT 06033

PMA NO.:	PQ0816NE
SUPPLEMENT No.:	ODA-TK-019
DATE:	March 3, 2016

Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Link, Flexible, Lever to Movable Panel	91E091-00DEC	Safran P/N: 91E091-00	Tests and Computations per 14 CFR § 21.303, Dwg No.: 91E091-00DEC, Rev.: B, Dated: 11/7/14, or later FAA approved revisions.	Airbus	A330-243
					A330-243F
					A330-341
					A330-342 A330-343
Cable-Flex Drive	491B1200036-001DEC	Middle River Aircraft Systems P/N: 491B1200036-001	Tests and Computations per 14 CFR § 21.303, Dwg No.: 491B1200036-001DEC, Rev.: C, Dated: 8/14/15, or later FAA approved revisions. <u>NOTE:</u> AMOC Required	General Electric	CF6-80C2B1
					CF6-80C2B2
					CF6-80C2B4
					CF6-80C2B6
					CF6-80C2B1F
					CF6-80C2B2F
					CF6-80C2B4F
					CF6-80C2B6F
					CF6-80C2B7F
					CF6-80C2B6FA
CF6-80C2B5F					
CF6-80C2B8F					
Cable-Flex Drive	491B1200036-002DEC	Middle River Aircraft Systems P/N: 491B1200036-002	Tests and Computations per 14 CFR § 21.303, Dwg No.: 491B1200036-002DEC, Rev.: C, Dated: 8/14/15, or later FAA approved revisions. <u>NOTE:</u> AMOC Required	General Electric	CF6-80C2B1
					CF6-80C2B2
					CF6-80C2B4
					CF6-80C2B6
					CF6-80C2B1F
					CF6-80C2B2F
					CF6-80C2B4F
					CF6-80C2B6F
					CF6-80C2B7F
					CF6-80C2B6FA
CF6-80C2B5F					
CF6-80C2B8F					

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Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Seal-Flexible Shaft	3238749-1DEC	Middle River Aircraft Systems P/N: 3238749-1	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3238749-1DEC, Rev.: C, Dated: 9/29/15, or later FAA approved revisions. <u>NOTE:</u> AMOC Required	General Electric	CF6-80C2B1 CF6-80C2B2 CF6-80C2B4 CF6-80C2B6 CF6-80C2B1F CF6-80C2B2F CF6-80C2B4F CF6-80C2B6F CF6-80C2B7F CF6-80C2B6FA CF6-80C2B5F CF6-80C2B8F

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Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Seal	3051-1048DEC	CFM International P/N: 3051-1048	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1048DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5 CFM56-5-A1/F CFM56-5A3 CFM56-5A4 CFM56-5A4/F CFM56-5A5 CFM56-5A5/F CFM56-5B1 CFM56-5B1/P CFM56-5B1/2P CFM56-5B2 CFM56-5B1/3 CFM56-5B2/3 CFM56-5B3/3 CFM56-5B3/3B1 CFM56-5B2/P CFM56-5B2/2P CFM56-5B3/P CFM56-5B3/P1 CFM56-5B4/3 CFM56-5B4/3B1 CFM56-5B5/3 CFM56-5B6/3 CFM56-5B3/2P CFM56-5B3/2P1 CFM56-5B4 CFM56-5B4/P CFM56-5B4/P1 CFM56-5B4/2P CFM56-5B4/2P1 CFM56-5B5 CFM56-5B5/P CFM56-5B6 CFM56-5B6/P CFM56-5B6/2P CFM56-5B7 CFM56-5B7/P CFM56-5B8/P CFM56-5B9/P CFM56-5B7/3 CFM56-5B8/3 CFM56-5B9/3 CFM56-5B9/2P

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Seal	3051-1048DEC	CFM International P/N: 3051-1048	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1048DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5C2 CFM56-5C2/4 CFM56-5C2/F CFM56-5C2/G CFM56-5C2/G4 CFM56-5C2/P CFM56-5C3/F CFM56-5C3/F4 CFM56-5C3/G CFM56-5C3/G4 CFM56-5C3/P CFM56-5C4 CFM56-5C4/P CFM56-5C4/1 CFM56-5C4/1P

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Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Seal, Trapezoidal	3051-1049DEC	CFM International P/N: 3051-1049	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1049DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5 CFM56-5-A1/F CFM56-5A3 CFM56-5A4 CFM56-5A4/F CFM56-5A5 CFM56-5A5/F CFM56-5B1 CFM56-5B1/P CFM56-5B1/2P CFM56-5B2 CFM56-5B1/3 CFM56-5B2/3 CFM56-5B3/3 CFM56-5B3/3B1 CFM56-5B2/P CFM56-5B2/2P CFM56-5B3/P CFM56-5B3/P1 CFM56-5B4/3 CFM56-5B4/3B1 CFM56-5B5/3 CFM56-5B6/3 CFM56-5B3/2P CFM56-5B3/2P1 CFM56-5B4 CFM56-5B4/P CFM56-5B4/P1 CFM56-5B4/2P CFM56-5B4/2P1 CFM56-5B5 CFM56-5B5/P CFM56-5B6 CFM56-5B6/P CFM56-5B6/2P CFM56-5B7 CFM56-5B7/P CFM56-5B8/P CFM56-5B9/P CFM56-5B7/3 CFM56-5B8/3 CFM56-5B9/3 CFM56-5B9/2P

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Seal, Trapezoidal	3051-1049DEC	CFM International P/N: 3051-1049	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1049DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5C2 CFM56-5C2/4 CFM56-5C2/F CFM56-5C2/G CFM56-5C2/G4 CFM56-5C2/P CFM56-5C3/F CFM56-5C3/G CFM56-5C3/G4 CFM56-5C3/P CFM56-5C4 CFM56-5C4/P CFM56-5C4/1 CFM56-5C4/1P

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Seal, Trapezoidal	3051-1050DEC	CFM International P/N: 3051-1050	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1050DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5 CFM56-5-A1/F CFM56-5A3 CFM56-5A4 CFM56-5A4/F CFM56-5A5 CFM56-5A5/F CFM56-5B1 CFM56-5B1/P CFM56-5B1/2P CFM56-5B2 CFM56-5B1/3 CFM56-5B2/3 CFM56-5B3/3 CFM56-5B3/3B1 CFM56-5B2/P CFM56-5B2/2P CFM56-5B3/P CFM56-5B3/P1 CFM56-5B4/3 CFM56-5B4/3B1 CFM56-5B5/3 CFM56-5B6/3 CFM56-5B3/2P CFM56-5B3/2P1 CFM56-5B4 CFM56-5B4/P CFM56-5B4/P1 CFM56-5B4/2P CFM56-5B4/2P1 CFM56-5B5 CFM56-5B5/P CFM56-5B6 CFM56-5B6/P CFM56-5B6/2P CFM56-5B7 CFM56-5B7/P CFM56-5B8/P CFM56-5B9/P CFM56-5B7/3 CFM56-5B8/3 CFM56-5B9/3 CFM56-5B9/2P

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Seal, Trapezoidal	3051-1050DEC	CFM International P/N: 3051-1050	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1050DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5C2 CFM56-5C2/4 CFM56-5C2/F CFM56-5C2/G CFM56-5C2/G4 CFM56-5C2/P CFM56-5C3/F CFM56-5C3/G CFM56-5C3/G4 CFM56-5C3/P CFM56-5C4 CFM56-5C4/P CFM56-5C4/1 CFM56-5C4/1P

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Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Seal	3051-1052DEC	CFM International P/N: 3051-1052	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1052DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5 CFM56-5-A1/F CFM56-5A3 CFM56-5A4 CFM56-5A4/F CFM56-5A5 CFM56-5A5/F CFM56-5B1 CFM56-5B1/P CFM56-5B1/2P CFM56-5B2 CFM56-5B1/3 CFM56-5B2/3 CFM56-5B3/3 CFM56-5B3/3B1 CFM56-5B2/P CFM56-5B2/2P CFM56-5B3/P CFM56-5B3/P1 CFM56-5B4/3 CFM56-5B4/3B1 CFM56-5B5/3 CFM56-5B6/3 CFM56-5B3/2P CFM56-5B3/2P1 CFM56-5B4 CFM56-5B4/P CFM56-5B4/P1 CFM56-5B4/2P CFM56-5B4/2P1 CFM56-5B5 CFM56-5B5/P CFM56-5B6 CFM56-5B6/P CFM56-5B6/2P CFM56-5B7 CFM56-5B7/P CFM56-5B8/P CFM56-5B9/P CFM56-5B7/3 CFM56-5B8/3 CFM56-5B9/3 CFM56-5B9/2P

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Seal	3051-1052DEC	CFM International P/N: 3051-1052	Tests and Computations per 14 CFR § 21.303, Dwg No.: 3051-1052DEC, Rev.: B, Dated: 1/5/16, or later FAA approved revisions.	CFM International	CFM56-5C2 CFM56-5C2/4 CFM56-5C2/F CFM56-5C2/G CFM56-5C2/G4 CFM56-5C2/P CFM56-5C3/F CFM56-5C3/F4 CFM56-5C3/G CFM56-5C3/G4 CFM56-5C3/P CFM56-5C4 CFM56-5C4/P CFM56-5C4/1 CFM56-5C4/1P

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Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Limiters Deflection	7066M26P02DEC	Middle River Aircraft Systems P/N: 7066M26P02	Tests and Computations per 14 CFR § 21.303, Dwg No.: 7066M26P02DEC, Rev.: C, Dated: 4/13/15, or later FAA approved revisions.	Airbus	A330-201 A330-202 A330-203 A330-301 A330-302 A330-303 A300, Model B2-1C A300, Model B4-2C A300, Model B2K-3C A300, Model B4-103 A300, Model B2-203 A300, Model B4-203 A310, Model 204 A310, Model 221 A310, Model 222 A310, Model 203 A310, Model 322 A310, Model 324 A310, Model 304 A310, Model 325 A300, Model B4-601 A300, Model B4-603 A300, Model B4-620 A300, Model B4-605R A300, Model B4-622R A300, Model F4-605R A300, Model F4-622R A300, Model B4-622 A300, Model C4-605R Variant F

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Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Limiter Deflection	7066M26P02DEC	Middle River Aircraft Systems P/N: 7066M26P02	Tests and Computations per 14 CFR § 21.303, Dwg No.: 7066M26P02DEC, Rev.: C, Dated: 4/13/15, or later FAA approved revisions.	Boeing	747-200B Series 747-200F Series 747-200C Series 747SR Series 747-300 Series 747-400 Series 747-400D Series 747-400F Series 767-200 Series 767-300 Series 767-300F Series DC-10-10 DC-10-30 DC-10-30F (KC-10A, KDC-10) DC-10-10F MD-11 MD-10-10F MD-10-30F

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Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Disk-Deflection Limiter	1314M14P01DEC	Middle River Aircraft Systems P/N: 1314M14P01	Tests and Computations per 14 CFR § 21.303, Dwg No.: 1314M14P01DEC, Rev.: C, Dated: 4/13/15, or later FAA approved revisions.	Airbus	A330-201 A330-202 A330-203 A330-301 A330-302 A330-303 A300, Model B2-1C A300, Model B4-2C A300, Model B2K-3C A300, Model B4-103 A300, Model B2-203 A300, Model B4-203 A310, Model 204 A310, Model 221 A310, Model 222 A310, Model 203 A310, Model 322 A310, Model 324 A310, Model 304 A310, Model 325 A300, Model B4-601 A300, Model B4-603 A300, Model B4-620 A300, Model B4-605R A300, Model B4-622R A300, Model F4-605R A300, Model F4-622R A300, Model B4-622 A300, Model C4-605R Variant F

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Disk-Deflection Limiter	1314M14P01DEC	Middle River Aircraft Systems P/N: 1314M14P01	Tests and Computations per 14 CFR § 21.303, Dwg No.: 1314M14P01DEC, Rev.: C, Dated: 4/13/15, or later FAA approved revisions.	Boeing	747-200B Series 747-200F Series 747-200C Series 747SR Series 747-300 Series 747-400 Series 747-400D Series 747-400F Series 767-200 Series 767-300 Series 767-300F Series DC-10-10 DC-10-30 DC-10-30F (KC-10A, KDC-10) DC-10-10F MD-11 MD-10-10F MD-10-30F
Bellows Assembly	88504DEC	Goodrich Aerostructures P/N: 88504	Tests and Computations per 14 CFR § 21.303, Dwg No.: 88504DEC, Rev.: D, Dated: 2/25/16, or later FAA approved revisions.	Airbus	A340-211 A340-212 A340-213 A340-311 A340-312 A340-313
Shaft-Flexible Drive	CH1543-0003-01DEC	Goodrich Actuation Systems P/N: CH1543-0003-01	Tests and Computations per 14 CFR § 21.303, Dwg No.: CH1543-0003-01DEC, Rev.: C, Dated: 8/31/15, or later FAA approved revisions.	Airbus	A319 Model -131 A319 Model -132 A319 Model -133 A320 Model -231 A320 Model -232 A320 Model -233 A321 Model -131 A321 Model -231 A321 Model -232

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Housing Assembly	DEC09-079-501	Pratt & Whitney P/N: 472592	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC09-079, Rev.: C, Dated: 2/21/06, or later FAA approved revisions.	Pratt & Whitney	JT8D-1 JT8D-1A JT8D-1B JT8D-7 JT8D-7A JT8D-7B JT8D-9 JT8D-9A JT8D-11 JT8D-15 JT8D-15A JT8D-17 JT8D-17A JT8D-17R JT8D-17AR JT8D-209 JT8D-217 JT8D-217A JT8D-217C JT8D-219
Nut-Special	DEC39-252-501	Honeywell P/N: 3236431-1	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC39-252, Rev.: F, Dated: 9/9/11, or later FAA approved revisions.	General Electric	CF6-80C2A1 CF6-80C2A2 CF6-80C2A3 CF6-80C2A5 CF6-80C2A5F CF6-80C2A8 CF6-80C2B1 CF6-80C2B1F CF6-80C2B2 CF6-80C2B2F CF6-80C2B4 CF6-80C2B4F CF6-80C2B5F CF6-80C2B6 CF6-80C2B6F CF6-80C2B6FA CF6-80C2B7F CF6-80C2B8F CF6-80C2D1F

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Sleeve-Outer	DEC39-263-501	Honeywell P/N: 3238329-23	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC39-263, Rev.: C, Dated: 8/02/10, or later FAA approved revisions.	General Electric	CF6-80C2A1 CF6-80C2A2 CF6-80C2A3 CF6-80C2A5 CF6-80C2A5F CF6-80C2A8 CF6-80C2B1 CF6-80C2B1F CF6-80C2B2 CF6-80C2B2F CF6-80C2B4 CF6-80C2B4F CF6-80C2B5F CF6-80C2B6 CF6-80C2B6F CF6-80C2B6FA CF6-80C2B7F CF6-80C2B8F CF6-80C2D1F
Gasket	DEC47-003-501	International Aero Engines P/N: 2A1226	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC47-003, Rev.: – , Dated: 6/11/04, or later FAA approved revisions.	International Aero Engines	V2500-A1 V2522-A5 V2524-A5 V2525-D5 V2527-A5 V2527E-A5 V2528-D5 V2530-A5 V2533-A5
Seal, Face	DEC56T178	Pratt & Whitney P/N: 56T178	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC56T178, Rev.: D, Dated: 11/12/14, or later FAA approved revisions.	Pratt & Whitney	PW4074 PW4074D PW4077 PW4077D PW4084D PW4090 PW4090D PW4098 PW4164 PW4168 PW4168A

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Retainer	DEC59-004-003	Boeing P/N: 314N3310-11	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC59-004, Rev.: G, Dated: 4/15/15, or later FAA approved revisions.	Boeing	757-200 Series
Fireseal, Fwd	DEC59-004-005	Boeing P/N: 314N3310-17	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC59-004, Rev.: G, Dated: 4/15/15, or later FAA approved revisions.	Boeing	757-200 Series
Fireseal, Aft	DEC59-004-007	Boeing P/N: 314N3310-16	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC59-004, Rev.: G, Dated: 4/15/15, or later FAA approved revisions.	Boeing	757-200 Series
Fireseal, Aft	DEC59-004-009	Boeing P/N: 314N3310-33	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC59-004, Rev.: G, Dated: 4/15/15, or later FAA approved revisions.	Boeing	757-200 Series
Fire Seal Assembly	DEC59-004-501	Boeing P/N: 314N3310-22	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC59-004, Rev.: G, Dated: 4/15/15, or later FAA approved revisions.	Boeing	757-200 Series

FAA – PARTS MANUFACTURER APPROVAL (PMA) SUPPLEMENT

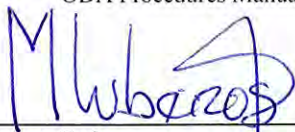
Turbine Kinetics, Inc.
 A HEICO Aerospace Company
 60 Sequin Drive
 Glastonbury, CT 06033

PMA NO.:	<u>PQ0816NE</u>
SUPPLEMENT No.:	<u>ODA-TK-019</u>
DATE:	<u>March 3, 2016</u>

Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Fire Seal Assembly	DEC59-004-503	Boeing P/N: 314N3310-32	Tests and Computations per 14 CFR § 21.303, Dwg No.: DEC59-004, Rev.: G, Dated: 4/15/15, or later FAA approved revisions.	Boeing	757-200 Series

----- End of Listing -----

NOTE: Minor design changes (reference 14 CFR part 21 §§ 21.319 and 21.619) and major design changes (reference 14 CFR part 21 §§ 21.319 and 21.619) to drawings and specifications must be accomplished in accordance with the FAA approved HEICO PMA ODA Procedures Manual.



 Marco Cuberos
 PMA ODA administrator
 HEICO Aerospace Corporation

 MARCH 3, 2016

Date

This Supplement is an attachment to FAA-PMA approval letter dated March 28, 2001



U.S. Department
of Transportation

Federal Aviation
Administration

February 22, 2005

Small Airplane Directorate
Atlanta Manufacturing Inspection
District Office
One Crown Center, Suite 475
1895 Phoenix Boulevard,
Atlanta, GA 30349

PMA No. PQ1469CE

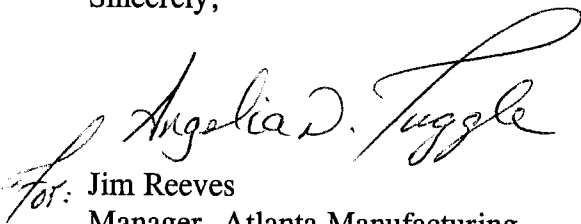
DEC Technologies, Inc.
501 Industrial Park Road
Piney Flats, TN 37686

FEDERAL AVIATION ADMINISTRATION-PARTS MANUFACTURER APPROVAL

In accordance with the provisions of 14 CFR part 21, Subpart K, we have found that the design data, based on Test & Computation submitted by DEC Technologies, Inc., with letters dated February 18, 2005, that meets the airworthiness requirements of the regulations applicable to the products on which the parts are to be installed. Additionally, it has been determined that DEC Technologies, Inc. has established the fabrication inspection system required by part 21 § 21.303(h) at 501 Industrial Park Road, Piney Flats, TN. Accordingly, Parts Manufacturer Approval (PMA) is hereby granted for production of the replacement parts listed in the enclosed Supplement No. 111.

You are reminded that the provisions of the Federal Aviation Regulations, noted in our PMA letter of approval dated October 17, 1995, also apply to the enclosed PMA Listing-Supplement No. 111.

Sincerely,

A handwritten signature in cursive script that reads "Jim Reeves".

For: Jim Reeves
Manager, Atlanta Manufacturing
Inspection District Office

Enclosure

FEDERAL AVIATION ADMINISTRATION-PARTS MANUFACTURER APPROVAL

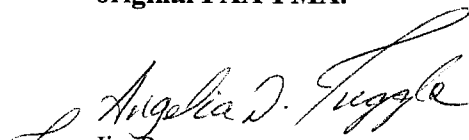
DEC Technologies, Inc.
501 Industrial Park Road
Piney Flats, TN 37686

PMA NO. PQ1469CE
SUPPLEMENT NO. 111
DATE: FEBRUARY 22, 2005

<u>Part Name</u>	<u>Part Number</u>	<u>Approved Replacement For Part Number</u>	<u>Approval Basis and Approved Design Data</u>	<u>Make Eligibility</u>	<u>Model Eligibility</u>
Bracket Actuator	DEC38-041-501	Grimes Aerospace Part Number: 46-2455-1 Used on Grimes Retractable Landing Light Assemblies Part Numbers: 45-0057-1, 45-0057-3, 45-0057-9 and 45-0057-11	Test and Computation per 14 CFR § 21.303(c)(4), Dwg. No: DEC38-041, Rev.: B, Date: 01/10/05, or later FAA Approved revisions	McDonnell Douglas	DC-9-14, -15, -15F, -21, -31, -32, -32F, -33F, -34, -34F, -41, -51, -81 (MD-81), -82 (MD-82), -83 (MD-83), -87 (MD-87), MD-88, MD-90
Valve-Fuel Pressurizing	DEC09-059-501	Pratt & Whitney 464392	Test and Computation per 14 CFR § 21.303(c)(4), Dwg. No: DEC09-059, Rev.: A, Date: 11/29/04, or later FAA Approved revisions	Boeing Pratt & Whitney	B717 JT8D-1, -1A, -1B, -7, -7A, -7B, -9, -9A, -11, -15, -15A, -17, -17A, -17R, -17AR, -209, -217, -217A, -217C, -219 ENGINES
Gasket	DEC47-003-501	IAE p/n 2A1226	Test and Computation per 14 CFR § 21.303(c)(4), Dwg. No: DEC47-003, Rev.: No. Date: 06/11/04, or later FAA Approved revisions	IAE	V2500-A1, V2522-A5, V2524-A5, V2525-D5, V2527-A5, V2527E-A5, V2528-D5, V2530-A5, V2533-A5
BRG, SLV Flanged TFE Lined	DEC47-004-501	IAE p/n 5A0783	Test and Computation per 14 CFR § 21.303(c)(4), Dwg. No: DEC47-004, Rev.: No. Date: 06/15/04, or later FAA Approved revisions	International Aero Engine	V2522-A5, V2524-A5, V2525-D5, V2527-A5, V2527E-A5, V2528-D5, V2530-A5, V2533-A5

-END OF LISTING-

NOTE: Minor design changes (reference 14 CFR Part 21 §§ 21.93 and 21.95) must be submitted in a manner as determined by the ACO. Major design changes (reference 14 CFR Part 21 §§ 21.93 and 21.97) to drawings and specifications are to be handled in the same manner as that for an original FAA-PMA.


By: Jim Reeves
Manager, Atlanta Manufacturing
Inspection District Office